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## **Advanced X-ray Imaging Satellite (AXIS)**

# **FACT SHEET**

Solar Array Wing

(1-Axis Gimballed, 2x)

#### Data Rate (MEV) Observatory: 12.6 Gb/day

**Key Instrument Modes** Target, ToO

Mass (MEV) Instrument: 1617 kg Spacecraft (dry): 1843 kg Observatory (wet): 3961 kg Observatory (dry): 3460 kg

Instrument: 939 W Spacecraft: 970 W Observatory: 1908 W

High-heritage spacecraft from an experienced vendor (Northrop Grumman)

Orbit-avg Power (MEV) Orbit Constraints Low-Earth Orbit Inclination = 5 degInsertion at 670 km Operate 610-670 km

**Concept:** LEO free flyer, Class C

Single Instrument: High

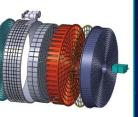
**Observatory Modes** 

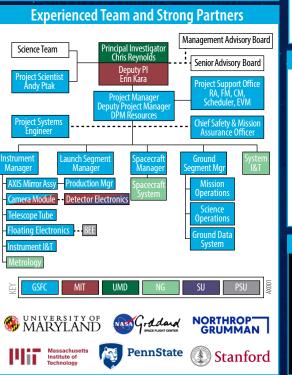
*Target:* 0.5-5ks exposures of fixed 450 arcmin<sup>2</sup> fields; slew to new target upon SAA passage or Earth occlusion. Transmit while observing *ToO*: 0.5-5ks exposure of target <2 hr after SOC receives alert. Transmit in <12 hr of alert

### Next Generation AXIS X-ray Mirror Assembly (XMA)

#### High-resolution, lightweight, silicon-segmentbased grazing incidence X-ray optic.

- 14,368 0.8mm segments
- Iridium-coated
- Packed into 264 modules
- Controlled to 22±0.25°C
- STOP analysis : 1.2 arcsec HPD (on-axis) CBE exceeds requirements





#### -FEE Rocket Engine Assembly resolution 9 m focal length Camera Assembly (3-22 N ΔV Thrusters) (hidden) (4x) X-ray imager (0.3-10 keV) Star Trackers (2) Spacecraft: Leverages Class / IRU Bench Camera Tube B LEOStar-3 bus (JPSS-2, Subassembly Landsat 8 & 9, ICESat-2) from Bus Optical Boresight Northrop-Grumman Propellant X-ray Mirror Assembly Tank (4x) FPA: Focal Plane Array Aperture Sunshade FFF: Front End Electronics (deployable) IRU: Inertial Reference Unit High-Speed AXIS X-ray Camera Module Small pixel, large-format, CCD Imaging Area low-noise, high-speed CCD Frame Store X-ray CCDs MCRC ASIC MIT/LL CCID-100 devices Operated at -90°C Aluminum Vacuum enclosure and warm Frame Store filter prevents contamination Cover FoV: 71 mm 27 arcmin Transient Alert Module finds Thermal Connection new sources real-time to -90°C Cold Finger AXIS Cost \$K (Fy23\$)

**Mission Overview** 

FPA Sunshade (fixed)

A CONTRACTOR OF THE OWNER		*				*					
Schedule											
		2025	2026	2027	2028	2029	2030	2031	2032	2033	2034 to 2036
NASA Phases	S <sup>O</sup> N <sup>D</sup> J <sup>F</sup> M <sup>A</sup> Phase A -	M <sup>J</sup> J <sup>A</sup> S <sup>O</sup> N <sup>J</sup> 12 Mo. [DS-(	GMo. Phase E	J <sup>F</sup> M <sup>A</sup> M <sup>J</sup> J <sup>A</sup> S <sup>O</sup> N <sup>D</sup> 8 - 19.5 Mo.	J <sup>F</sup> M <sup>A</sup> M <sup>J</sup> J <sup>A</sup> S <sup>U</sup> N <sup>D</sup> .	Phase C - 42 M	j <sup>r</sup> M <sup>A</sup> M <sup>J</sup> J <sup>A</sup> S <sup>O</sup> N <sup>L</sup> No.	Phase D	- 14 Mo.	J <sup>F</sup> M <sup>A</sup> M <sup>J</sup> J <sup>A</sup> S <sup>O</sup> N Phase E - Basel	ine 60 Mo. >>> [F]
NASA HQ Milestones	<b>\</b> KDP-A 10/1	Site Vi	KDP-B 4/2 sit	KDP-C 11	/18			KDP-D 5/14	KDP-E 6/9		KDP-F 7/23
AXIS Mission Milestones			SRR 9/28 ATP 4/2	PDR 9/17	CDR 10/20	)		PER 6/24 SIR 4/29 FRR/I	Launch 6/23 RR 5/13 PLAR 7/2:	8	

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